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# **International Standard Problem on containment thermal-hydraulics ISP47 - STEP1 - Results from the MISTRA exercise**

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## **ABSTRACT**

The understanding of hydrogen distribution during severe accidents in a nuclear reactor containment is still an open issue. Several containment thermal-hydraulics international standard problems (ISP) have been conducted to address this topic. However the predictions made by the available Lumped Parameter or CFD computer codes were generally not satisfactory. Therefore a new exercise was launched in 1999 using new state-of-the-art experimental facilities TOSQAN, MISTRA and ThAI that included sophisticated 3D instrumentation and well controlled boundary conditions. Predictive capabilities of important and still uncertain phenomena such as wall condensation, natural circulation and gas stratification are assessed. In addition, comparison between LP and CFD codes and assessment of the capability of CFD codes to deal with scaling effects are performed. This article reports on the part of the exercise which concerns the MISTRA facility including experimental results and blind benchmark exercises.

## **KEYWORDS**

CONTAINMENT THERMAL-HYDRAULICS, MISTRA, ISP47, BENCHMARK EXERCISE

# 1 INTRODUCTION

During the course of a severe accident in a Pressurised Water Reactor (PWR), large amounts of hydrogen can be released into the containment (Haste et al., 1995). As a consequence, the integrity of this third barrier can be compromised by severe combustion events. As these hydrogen combustion processes involving slow and fast deflagrations for example depend on narrow gas concentration range, accurate gas distribution calculations are needed. An international consensus (OECD/NEA, 1999) related to previous international standard problems (ISP) (Karwat, 1989, 1993, NUPEC, 1994) concluded that detailed knowledge of the containment thermal-hydraulics is necessary for reliable predictions of pressure, temperature and gas concentration fields. The main objective of the present OECD International Standard Problem (ISP47) is to contribute to assess the capability of Lumped Parameter (LP) and CFD computer codes to predict the hydrogen distribution under severe accident conditions in a Light Water Reactor. The objectives of this new exercise are:

- to cover phenomena important and still uncertain for containment thermal-hydraulics such as wall condensation, natural circulation, atmosphere stratification, turbulent diffusion as well as interaction between them,
- to bring together users of LP and CFD codes and
- to assess the capability of CFD codes to deal with scaling effects.

This ISP has been mainly conducted in two steps using newly available facilities, each with sophisticated instrumentation:

- Step 1 is dedicated to the validation of refined models and scaling effect studies using TOSQAN and MISTRA facilities. Wall condensation and buoyancy are addressed in a rather simple geometry. Simultaneously, the validation of the interactions of phenomena, such as condensation/stratification, turbulence/buoyancy, etc., including the effect of scale-up allowed by the larger scale of MISTRA are investigated in a simple flow pattern.
- Step 2 addresses the validation of computer codes in a multi-compartmented geometry with asymmetric injections in the ThAI facility. This configuration is more representative of the real plant complexity.

Step 1 is divided into 2 separate phases: Phase A dealing with Air/Steam steady states in TOSQAN and MISTRA and Phase B looking at the effect of a Helium tran-

sient injection prior to Air/Steam/Helium steady states in TOSQAN and MISTRA facilities. The ISP was organised in an "open" way for the TOSQAN calculations allowing for model adjustments and in a "blind" way for the MISTRA calculations. It should be mentioned that some of the participants had already performed calculations of a MISTRA experiment during the MICOCO benchmark organised by CEA (Blumenfeld and Paillère, 2003). This may be important for the blind results of Phase A.

This article synthesises the MISTRA exercise involving experimental results and benchmark calculations. In particular, it mainly addresses the achievement of experimental data suitable for LP/CFD code benchmarking, the simulation of flow pattern close to a vertical injection, the distribution of condensation and the behaviour of saturation along a vertical condensing wall and the effect of helium addition on that condensation process. The results of the whole ISP will be published at the completion of the exercise. In the first section, we describe MISTRA facility and the initial and boundary conditions relevant to the present exercise. The second section is organised around comparisons between experimental and computed results for global and local variables. Some interesting transient results occurring between the two phases are then described and discussed in the third section. Conclusions follow.

## **2 Description of MISTRA facility and initial and boundary conditions related to ISP47**

The MISTRA facility is part of the CEA programme related to severe accidents occurring in nuclear reactors and is focused on containment thermal-hydraulics and hydrogen safety. Several papers already provided a description of the MISTRA facility (Studer et al., 2003, Tkatschenko et al., 2005) and in the present article we focus on the geometrical description and the associated initial and boundary conditions related to the ISP47 exercise.

### *2.1 MISTRA facility and associated instrumentation*

The MISTRA facility is a stainless steel vessel of 99.5 m<sup>3</sup> (Fig. 1). The internal diameter (4.25 m) and the height (7.3 m) were chosen to scale to a typical French PWR containment with a linear length scale ratio of 0.1. The pressure vessel com-

prises 2 shells, a flat cap and a bottom, which are attached with twin flanges. The external part of the vessel (about 118 m<sup>2</sup>) itself is not temperature-regulated but thermally insulated with 20 cm of rock wool.

Three independent thermally driven walls, hereafter called condensers are inserted inside the containment close to the external wall: 26.2 m<sup>2</sup> for the lower condenser and 21.4 m<sup>2</sup> for the middle and the upper condenser. A so called “dead volume” behind the condensers exists and spurious condensation may occur during experiments (due to heat losses). The condensers are specially designed to ensure well controlled boundary conditions (+/- 1°C). The external part is insulated with 2 cm of synthetic foam and viewing windows are installed for laser measurements. Condensates are collected in gutters: one per condenser and three circuits for the spurious condensation (containment walls, insulated part of the condensers and sump). The viewing windows installed in the middle condenser are used to provide Laser Doppler Velocimetry measurements along a radial profile. The containment atmosphere is seeded by SiC particles but the presence of fog at least close to the condensers (middle and lower) during the Phase B steady-state did not allow this measurement for the second steady state.

A diffusion cone fitted with a removable cap is designed for gas injection (200 mm of injection diameter). A porous medium is used to ensure flat velocity profiles at the injection nozzle and the injection device is set-up in the bottom along the central axis (Fig. 2). The flow is injected at a given rate controlled and measured using sonic nozzles that ensure a constant value independent of the downstream operating conditions. The injection temperature is usually measured beyond the porous medium, inside the diffusion cone (1285 mm elevation). During the 2003 dismantling phase, leakages inside the diffusion cone were detected due to failure of the main seal of the porous medium. So, extra turbulence was created at the injection level for the first set of experiments performed in 2002. For the 2004 tests series, the porous medium was welded on the injection device preventing leaks for the new trial runs. The instrumentation location is displayed in Fig. 3 for the main vertical half-planes (345° azimuthal angle). Thermocouples and gas sampling tubes are installed on this grid and the maximum distance between two sensors is less than 1 meter axially and 0.5 m radially. Two others half planes (105° and 225°) are lightly instrumented mainly to check the symmetry of the flow.

## 2.2 Initial and boundary conditions

The simplified test sequence related to the ISP47 exercise is made up of four successive phases:

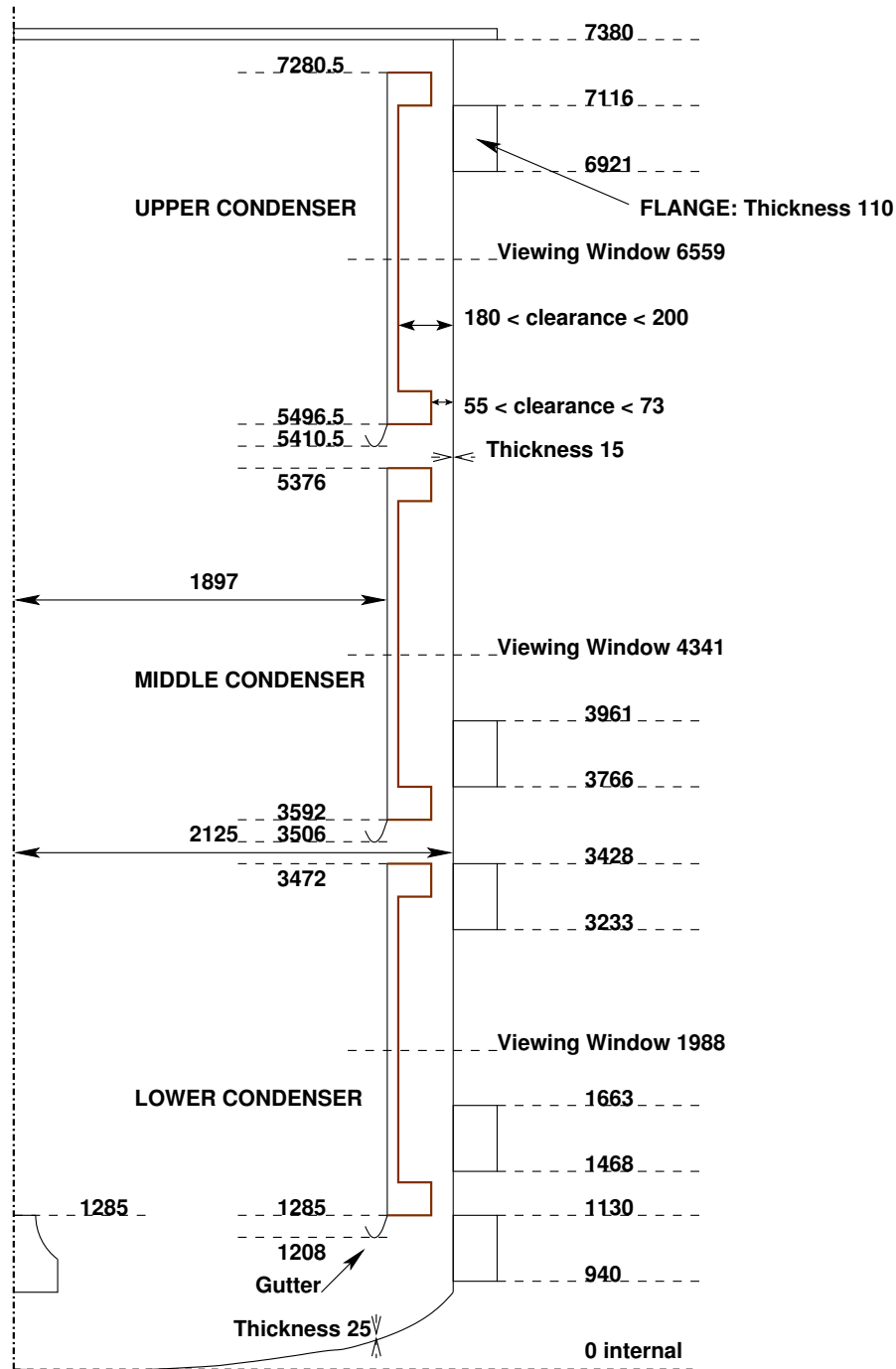


Fig. 1. Schematic view of the MISTRA facility, symmetry half plane (dimensions are given in mm)

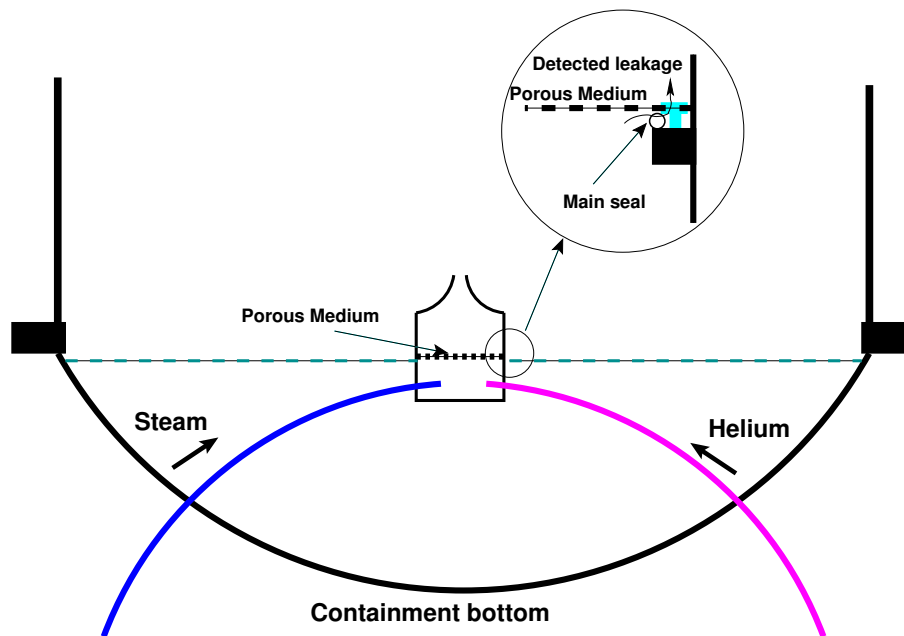


Fig. 2. Schematic view of the injection device used in the MISTRA facility

- (1) **Preheating phase:** superheated steam injection into the facility initially at room temperature and pressure. This is a process phase mainly used to heat up the steel structures and the results of this phase are not reported and discussed in the present document.
- (2) **Air/Steam steady-state (Phase A)** defined from the balance between the injected and condensed mass flows (130 g/s) ensuring the stability of all the parameters: pressure, temperature and gas concentrations.
- (3) **Air/Steam/Helium transient** mass flow of helium (simulating hydrogen) is added to the main steam mass flow at a rate of 10 g/s for half an hour.
- (4) **Air/Steam/Helium steady-state (Phase B)** with the same definition and boundary conditions as for Phase A.

The test was run ten times to check for repeatability (6 times in 2002 and 4 times in 2004) and also to allow gas concentration measurements. Initial and boundary conditions are given in terms of mean values and the given uncertainty includes systematic and repeatability errors. Initial conditions are only needed to specify the air density which is constant during the test. A mean value of  $1.195 \text{ kg/m}^3$  with an uncertainty of about  $0.02 \text{ kg/m}^3$  is obtained for the tests. The superheated steam mass flow is injected at a constant rate during the test:  $130.1 \text{ g/s}$  with an estimated uncertainty of  $\pm 3 \text{ g/s}$ . The helium mass flow is injected at a rate of  $10.16 \text{ g/s}$  (with an estimated uncertainty of  $\pm 0.35 \text{ g/s}$ ) for a duration of 1826 seconds. By supposing any leakage, this leads to about 118.9 kg of air and 18.6 kg of helium

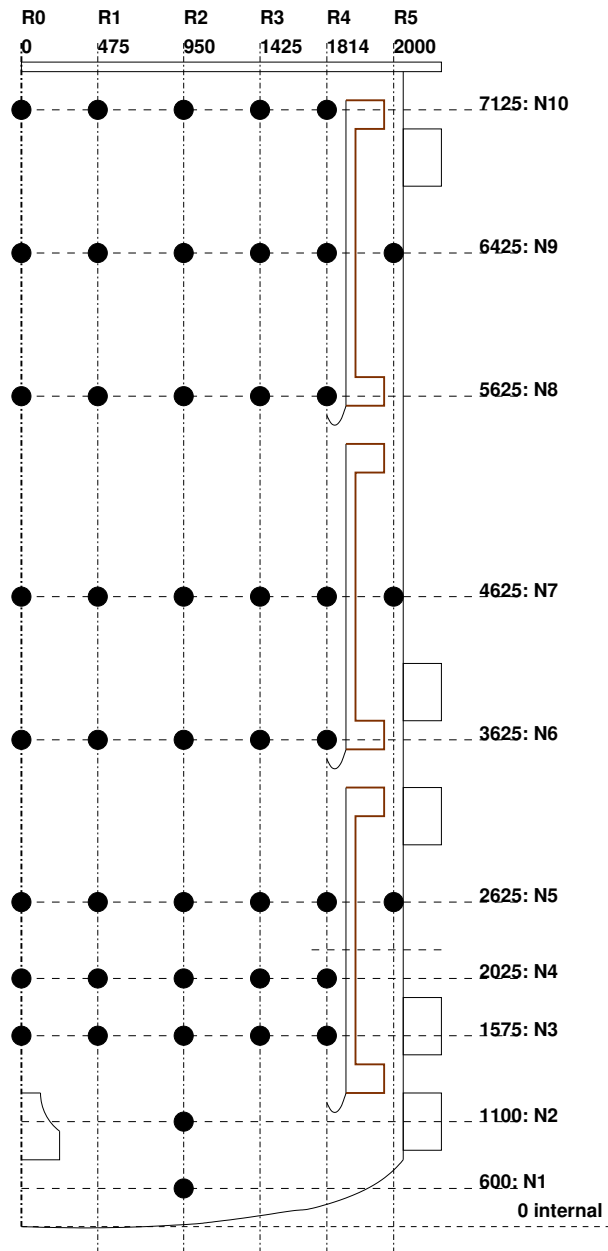


Fig. 3. Main instrumented vertical plane of the MISTRA facility (dimensions in mm)

inside the facility.

The injection temperature at the steady state is about 198.1°C for Phase A and 201.6°C for Phase B with an uncertainty of about +/- 2.3°C. A transient behaviour is measured (Fig.4) during the helium injection phase (HTI). During Phase A steady-state, steam enters the helium injection pipe (Fig.2) and condenses outside the facility where the pipe is not heated. A water plug is created inside the pipe. Then, hot helium is injected and has to heat up the pipe and to evaporate the water plug.



Improvements during the 2003 upgrading phase (purge of the water plug prior to helium injection and additional electrical heating of the helium injection pipe outside the facility) have reduced this temperature decrease. Finally, it was shown that these different transients conditions have no impact on the transient behaviour of the condensation described below.

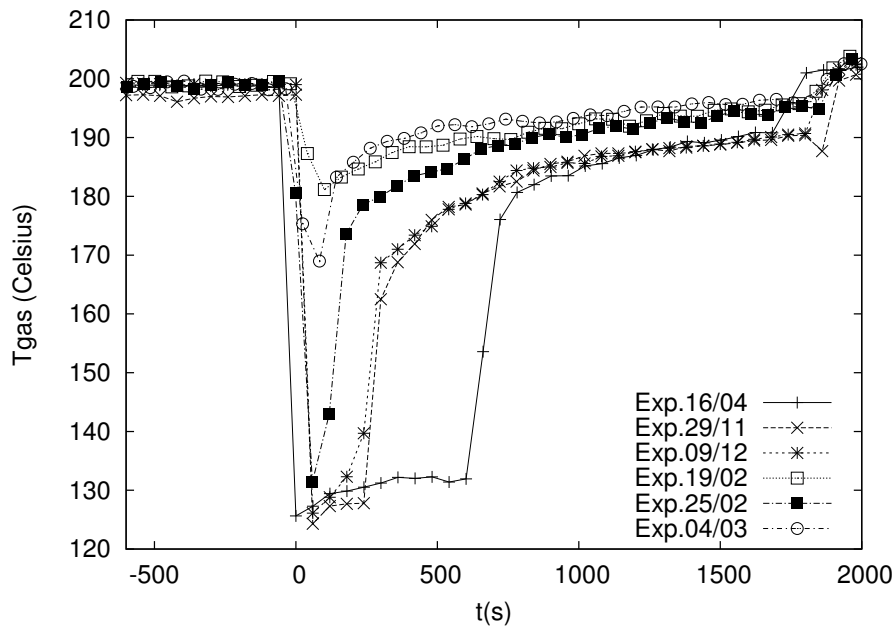


Fig. 4. ISP47: injection temperature of the gaseous mixture (Steam and Helium) during the transient addition of helium (2002 tests series: 16/04, 29/11 and 09/12 - 2004 tests series 19/02, 25/02 and 04/03)

For the condensers, the specified temperature is  $115^{\circ}\text{C}$  for each condenser and the achieved conditions are  $115.2^{\circ}\text{C}$  for the lower condenser,  $114.6^{\circ}\text{C}$  for the middle condenser and  $115.3^{\circ}\text{C}$  for the upper condenser with a thermocouple uncertainty of about  $\pm 0.8^{\circ}\text{C}$ . This demonstrates controlled boundary conditions in the MISTRA experiments.

Spurious condensation along the external steel walls was specified for the benchmark exercise (12% of the injected mass flow rate ie. 15.6 g/s) and the measured values corresponds to 17.9 g/s for Phase B and between 15.2 to 17.5 g/s for Phase A (distributed as follow: 40% on the containment walls, 40% on the sump and 20% on the insulated material on the external side of the condensers). This spurious condensation results in heat loss. The major part of these heat losses occurs at the sump due to penetrations and supporting structures. For the walls of the dead volume behind the condensers, heat losses of about 14 kW correspond to an estimated heat

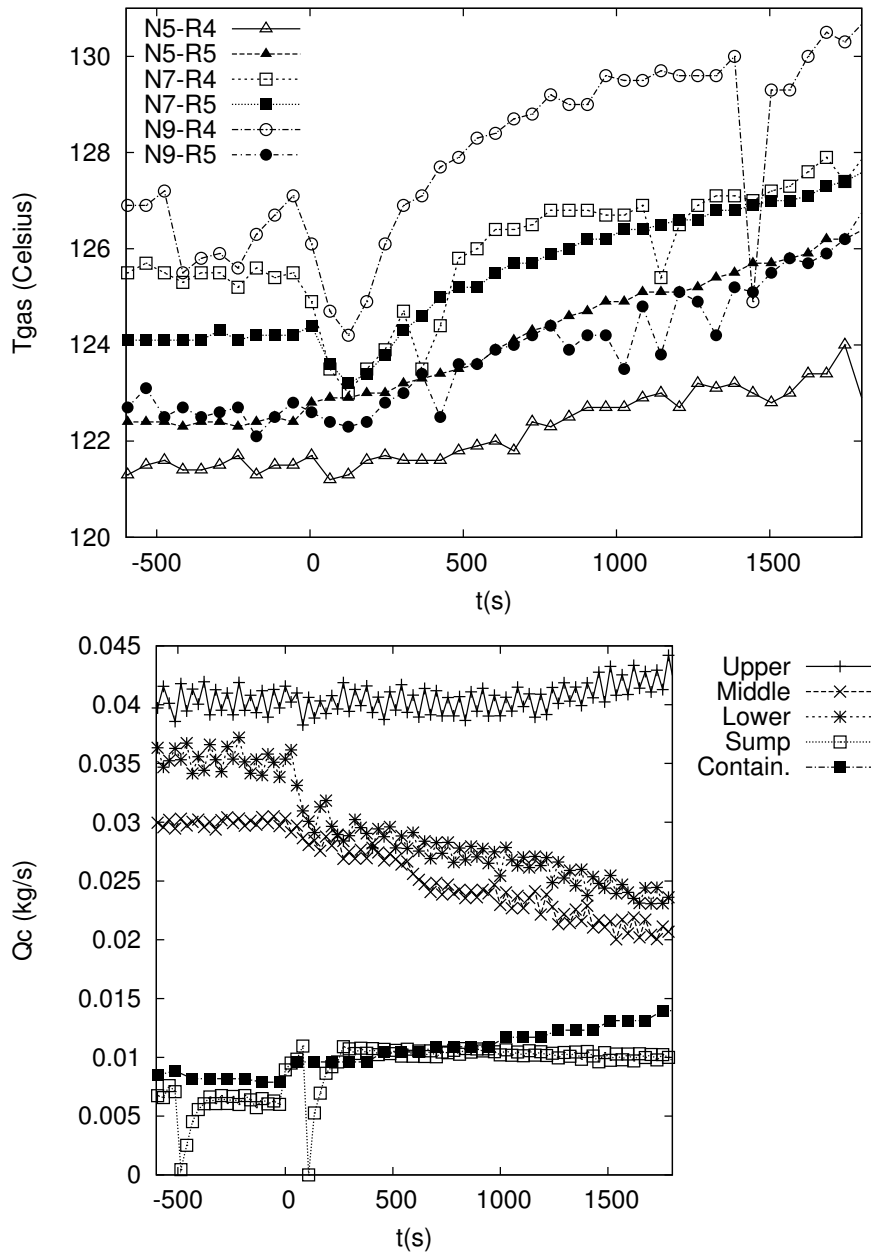


Fig. 5. ISP47 - behaviour of the “dead” volume during Helium transient addition - 2002 tests series - time zero corresponds to the start of helium addition (Top: gas temperature - Bottom: condensation flow)

conductivity of the insulation material of 0.3 W/m/K. Similar value (0.1 W/m/K) has also been derived for the ThAI facility. For the benchmark different strategies were proposed depending on the code capabilities. One strategy is to remove the spurious condensation mass flow from the injected mass flow. Alternatively, one can compute the thermal behaviour of the steel structure using an external heat exchange coefficient of 3-4 W/m<sup>2</sup>/K with a hall at a constant temperature of 20°C.

This assumes that the heat losses are homogeneously distributed. These initial and boundary conditions are summarized in Fig. 6.

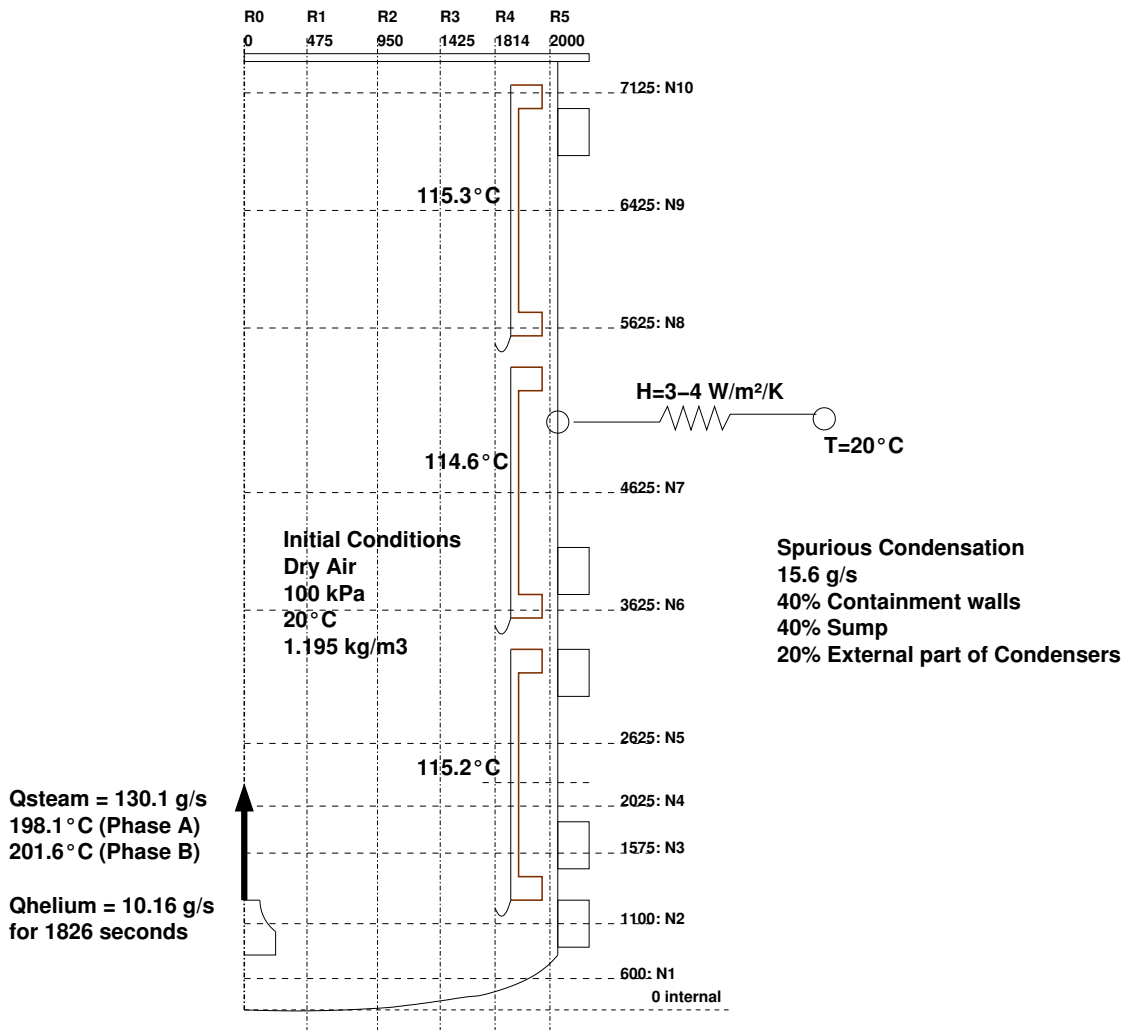


Fig. 6. ISP47: initial and boundary conditions

Examples of the atmospheric conditions in the dead volume are presented in Fig. 5 (top). First, the decrease of the injection temperature (Fig. 4) is reported in front of the condensers (R4 axis) and behind them (R5 axis). Then, the thermal transient is almost the same in front of and behind the condensers. Finally, the condensation mass flow gradually increases. So, from the available measurements, it seems that the dead volume follow the main gaseous volume.

Improvement of the mass balance (injection/condensation) was performed during the dismantling phase. The 2002 tests series lead to about 1 to 7 % difference and this reduces to less than 1 % for the 2004 tests series.

### 3 Participants in the benchmark exercise

Most of the institutions and computer codes involved in containment thermal-hydraulics analysis in Nuclear Reactor safety participated in this benchmark exercise (Table 1). 18 contributions used 6 CFD codes (5 in-house and only 1 commercial) and 8 LP codes. Some differences remain in the versions used for the calculations. Scaling-up between the different facilities involved in this ISP benchmark was identified as an important topic and the main characteristics of the nodalisation used are reported in Table 2 (the length scale for CFD codes and the number of nodes for lumped parameter codes). Some participants submitted different contributions: NUPEC with a 2D model and a partial 3D models (1/4 of the MISTRA Facility), CEA with a LP, a 2D and a full 3D model.

The Lumped Parameter models used between 1 and 52 nodes to model the MISTRA facility. Radially, 3 to 5 levels were adopted depending on the choices regarding the volume behind the condensers and the radial discretisation of the main gaseous volume into 3 zones (upward flow due to the jet, stagnant zone in the middle and downward zone near the condensers) or fewer. Axially, two different approaches can be distinguished: a coarse grid involving mainly 4 levels corresponding to the 3 condensers and the bottom; a finer grid using between 7 to 14 levels. One important parameter when using lumped parameter codes is the friction or form loss coefficient used to model the atmospheric junctions between compartments (approximation of the momentum equation). GRS used the same form loss coefficient in vertical and horizontal directions ( $K_{form} = 0.3$ ) except for horizontal paths between the inner room and the dead volume where  $K_{form} = 1.0$  is used. CEA used different coefficients for the horizontal or vertical junctions (two times higher in the horizontal direction). IRSN adjusted the horizontal coefficients between 1.0 and 10.0 to numerically obtain only one convective loop. For phase B, it used the same approach as for CEA. LEI used constant coefficients in both directions. Rules should be clearly elaborate at the end of the ISP regarding this point and especially guidelines for extrapolation at the reactor scale.

In 2D, some participants used coarse grids (AECL and NAI) but with the objective of being consistent with the real containment analysis. An intermediate number of nodes was used by FZK, NUPEC, NRG and CEA. PSI used a relatively fine grid to model the MISTRA facility. In 3D, the two grids are equivalent because NUPEC only modelled 1/4 of the MISTRA facility. Grid convergence of CFD calculations

Table 1  
 Summary of ISP47 benchmark participants (Vdev: version under development)

Number	Organisation	Country	Computer code	Version	Category
1	AECL	Canada	GOTHIC (George et al., 1999)	6.1bp2	CFD
2	UJV	Czeck Republic	MELCOR (Gaunt et al., 2000)	1.8.5(A)QZ(+Patch002)	LP
3	IRSN	France	TONUS (Studer et al., 2003) ASTEC (H.J. Allelein et al., 1999)	V2002.2 V0.4	LP
4	FZK	Germany	GASFLOW II (Travis et al., 1999)	V2.2.4.21	CFD
5	GRS	Germany	COCOSYS (Klein-Hessling et al., 2000) ASTEC (Van Dorselaere et al., 2005)	V2.0 V1.0	LP
6	VEIKI	Hungary	ASTEC (Van Dorselaere et al., 2005)	V1.0	LP
7	University of Pisa	Italy	FUMO (Manfredini et al., 2002)	Vdev	LP
8	NUPEC	Japan	DEFINE (Ishida et al., 2001)	Vdev	CFD
9	NRG	Netherlands	CFX (CFX4.4, 2001)	4.4	CFD
10	IJS	Slovenia	CONTAIN (Murata et al., 1997)	2.0	LP
11	STUDVISK	Sweden	MELCOR (Gaunt et al., 2000)	1.8.5	LP
12	PSI	Switzerland	CFX (CFX4.3, 1999)	4.3	CFD
13	NAI	U.S.A.	GOTHIC (George et al., 2003)	7.2dev	LP and CFD
14	LEI	Lithuania	COCOSYS (Klein-Hessling et al., 2000)	2.0	LP
15	IPPE	Russia	KUPOL-M (Soloviev et al., 2003)	1.10	LP and CFD
16	CEA	France	TONUS (Studer et al., 2003)	Vdev	LP and CFD

Identification	Number of nodes	Radial length in cm (Min/Max)	Axial length in cm (Min/Max)
GOT2D1	228	2.5 / 27.4	12.0 / 45.1
MELLP2	12	3 levels max	4 levels
TONLP3	13	4 levels max	4 levels
ASTLP3	13	4 levels max	4 levels
GAS2D4	3528	5.0 / 10.5	2.75 / 7.0
COCLP5	45	4 levels max	12 levels
ASTLP6	44	4 levels	11 levels
FUMLP7	13	4 levels max	5 levels
DEF2D8	972	8.2 / 8.2	10 / 33
DEF3D8	19950	8.2 / 8.2	10 / 33
CFX2D9	5717	2 / 13	2 / 13
CONLP10	1	-	-
MELLP11	18	3 levels max	7 levels
CFX2D12	7116	1.5 / 11.8	1.4 / 6.1
GOT2D13	437	5 / 10	37.7 / 42.8
COCLP14	19	3 levels max	7 levels
KUPLP15	52	5 levels max	14 levels max
KUP2D15	1428	5.5 / 10.6	16 / 21.2
ASTLP5	45	4 levels max	12 levels
TONLP16	31	5 levels max	9 levels
TON2D16	3146	4 / 10	7.5 / 10
TON3D16	85123 (fluid) 34779 (solid)	3.3 / 10	10 / 10

Table 2

Geometrical discretisation of MISTRA facility (max: means the maximum number of nodes in a specific radial or axial axis)

is often mentioned as a necessary condition. Nevertheless, in the present exercise wall condensation in CFD is modelled by using empirical heat and mass transfer correlations. So, characteristic length scales are implicitly introduced in these correlations that are not compatible with grid convergence analysis.

Regarding the models used by the different participants, only some characteristics related to the present exercise are detailed:

- **Equations of state for steam:** most of the participants used real gas equations for steam except for the CFD contributions of NRG, PSI and CEA where a perfect gas hypothesis was made. Such hypothesis is relevant for superheated state conditions and can lead to supersaturated conditions especially close to the condensers (see comments below).
- **filmwise steam condensation** models implemented in most of the codes are based on global correlation (Uchida et al., 1965) or heat and mass transfer (HMT) analogy (Gido and Koestel, 1982, Collier, 1974) except for GASFLOW code where wall functions are implemented. Regarding commercial CFD codes, correlations based on HMT analogy were implemented for the present exercise.
- **fog formation or homogeneous condensation in CFD:** only GOTHIC and GASFLOW codes have homogeneous condensation models.
- **turbulence models in CFD:** so called standard  $k-\varepsilon$  models are implemented in the different CFD codes except in TONUS code where an algebraic model based on a single and spatially constant mixing length is used.

## 4 Steady state results

### 4.1 Pressure and mean temperature

Accurate predictions of the pressure during a severe accident scenario may be the most important requirement for a severe accident computer code. For the two phases A and B, the results are displayed in Fig. 7 (the experimental uncertainties are estimated of +/- 0.03 bar). Three participants (MELLP2, MELLP11 and CONLP10) have computed pressure with more than 10% overestimation (Phases A and B). Post-test calculations for the contribution MELLP2 have shown that this overestimation is related to the set of correlation used by MELCOR code to solve heat and mass transfer along walls. Generally, most of the participants tend to un-

derestimate the pressure (about 0.2 bar) at phase B steady-state and overestimate (0.1 bar) at phase A steady-state and no clear distinction can be made between LP or CFD contributions. Differences are related to the steam mass content and so depend on the modelling of steam mass transfer at the wall. If we assume that the steam mass flux at the film interface is expressed by

$$\Phi_{steam,int} = k_m \frac{\rho_{steam,bulk} - \rho_{steam,int}}{1 - X_{steam,int}}, \quad (1)$$

where  $X$  denotes the steam molar fraction,  $k_m$  is the mass transfer coefficient and  $\rho$  the density, this flux must be the same for Phases A and B (same steam injection conditions). Since the interface conditions (subscript *int*) are only constrained by the condenser temperature and the presence of helium decreases the mass transfer coefficient, the mass flux may be kept constant only by increasing the steam content in the bulk. To illustrate this point, the differences for example, in the two GOTHIC contributions (pressure of 5.13 bar for GOT2D1 and 5.77 bar for GOT2D13) are related to the above phenomenon. The mean temperature differences can only lead to about only 1% pressure increase and different wall condensation modelling are used: MDML model which is a kind of HMT analogy including the generation of mist in the boundary layer in GOT2D13 contribution and maximum value between Uchida (Uchida et al., 1965) and Gido-Koestel (Gido and Koestel, 1982) correlations in GOT2D1 contribution. For the COCOSYS/ASTEC code (5 contributions), the computed pressure for Phase B is between 5.0 and 5.32 bars (+/- 3%). Detailed explanations of the differences in steam mass content would involve a detailed examination of the wall condensation modelling but it is beyond the scope of the present benchmark exercise.

Mean temperature prediction at the steady-state can yield an interesting information but evaluation of such a temperature is difficult experimentally (intensive variable based on a limited number of thermocouples). Mean temperatures are derived by using the thermocouples along the R2 vertical axis and by weighting their measurements by the thermocouple distance. The experimental results give 124.4°C for Phase A and 128.2°C for Phase B. By using global correlations such as Uchida for example, this gas temperature increase is related to the decrease of the total heat transfer coefficient (condensation and convection). As for the simulations, the mean gas temperature ranges between 121°C and 135°C for Phase B and the same deviations are obtained for Phase A. Gas temperatures are mainly controlled by the convective part of the heat transfer along the condensers (see appendix) and within the present benchmark exercise, correlations between these two variables were not



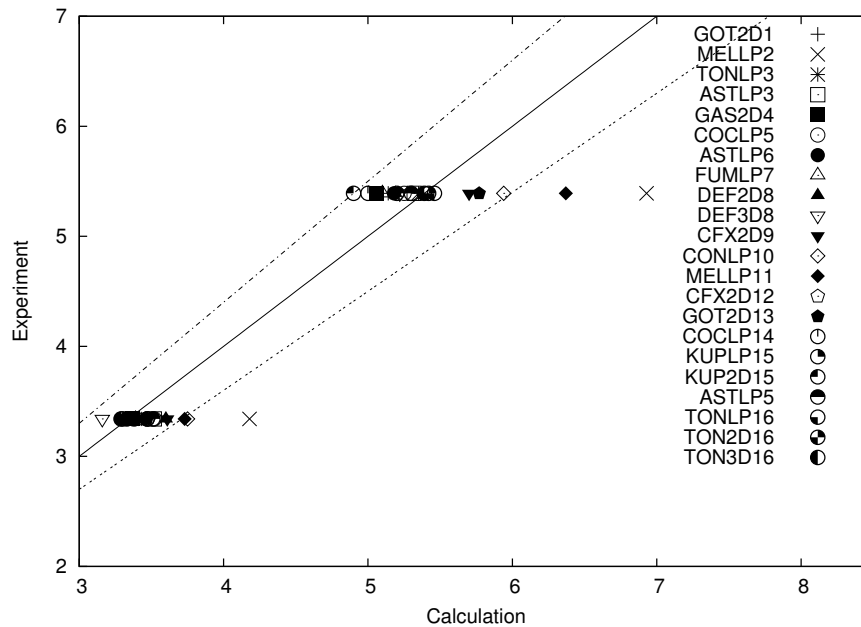


Fig. 7. Pressure (in bar) for the two steady-states A and B (dotted lines correspond to 10% deviation) seen.

#### 4.2 Condensation distribution along condensers

Condensation distribution along the condensers is experimentally influenced by the injection conditions (Table 3). The improvements regarding the injection device (described in 2.1) have introduced some differences in the condensation distribution mainly regarding Phase A steady-state. The steam condensation flux increases on the upper condenser and a reduction of this condensation flux is measured between the upper and the lower condensers (decrease of the steam mass fraction in the downward flow). Phase B results clearly show the increase of the contribution on the upper condenser compared to the middle condenser. Compared to the Phase A results, the differences regarding the two tests series (2002 and 2004) are not so important. For the Phase B, the condensation distribution along the condensers is mainly controlled by the condenser themselves rather than the mixing process in the upward flow.

Comparisons between experiments and calculation require an additional variable. Spurious condensation has been introduced by the participants in different ways. Some participants subtracted this variable from the injection mass flow rate, others

Identification	Phase A 2002	Phase A 2004	Phase B 2002	Phase B 2004
Upper condenser	1.949	2.233	2.355	2.645
Middle condenser	1.491	1.836	1.093	1.266
Lower condenser	1.401	1.095	1.225	1.134

Table 3

Steam condensation fluxes ( $g/s/m^2$ ) along the three condensers (upper condenser  $21.4m^2$ , middle condenser  $21.4m^2$  and lower condenser  $26.2m^2$  and the difference up to the mass balance is part of the spurious condensation)

do not take it into account or model it using the thermal behaviour of the external vessel. Now, we consider the part represented by each condenser (Fig. 8 and 9) in the total condensed mass flow along the three condensers (about 88% of the injected mass flow rate). ASTLP6 contributions (not reported in Fig. 8 and 9) have shown a special behaviour with condensation occurring only on the lower condenser. Post-test calculations have shown that this specific flow pattern is due to a absence of specific nodalisation close to the injection. In a LP code, orientation of the injection (vertical in the MISTRA facility) is obtained by driving the injected flow upwards (enough detailed nodalisation, pressure drop coefficient or absence of horizontal atmospheric junction).

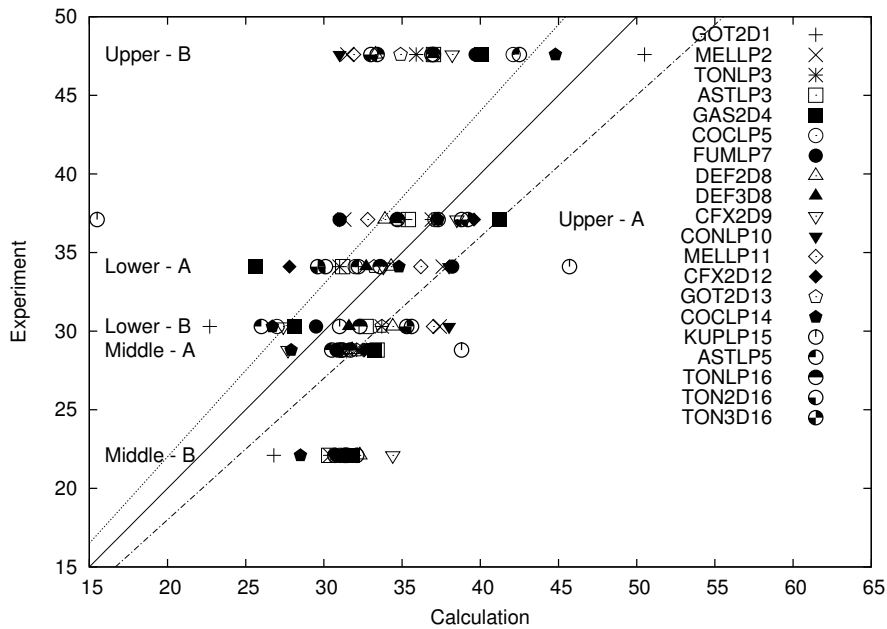


Fig. 8. Condensation distribution (percentage) for the two steady-states A and B - 2002 tests series (dotted lines correspond to 10% deviation)

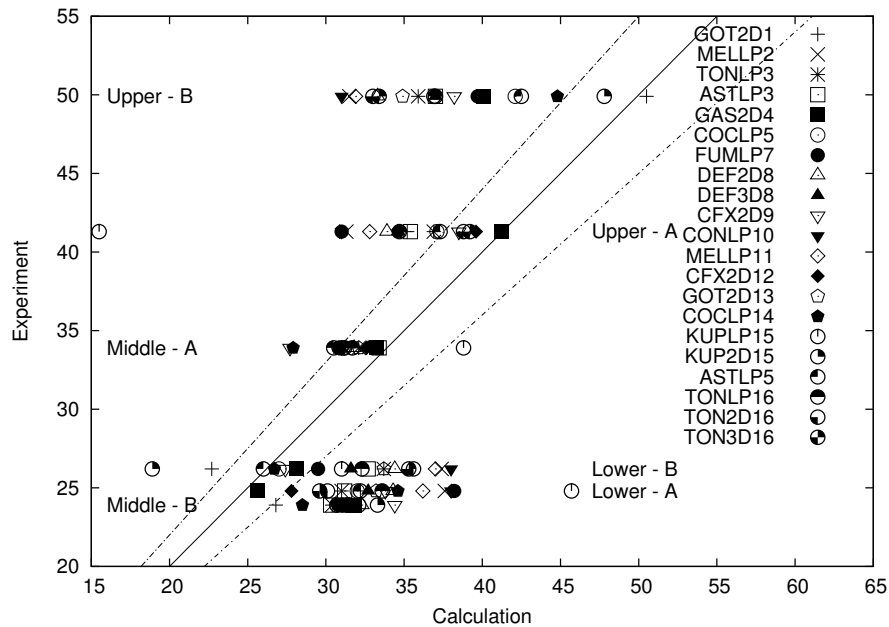


Fig. 9. Condensation distribution (percentage) for the two steady-states A and B - 2004 tests series (dotted lines correspond to 10% deviation)

Looking at the other contributions, the range of the computed values is the widest for the upper condenser. Several reasons may be pointed out: the flow has to turn before the development of the boundary layers, higher steam content in the upper part of the facility and condensation in the dead volume. Condensation models are probably not adequate for such situations because these correlations address diffusion of steam through non-condensable gases in a fully developed boundary layer. This mainly leads to an underestimation of the condensation mass flow rate. For phase A, the LP and CFD codes run on coarse grids or which use mixing length turbulence models lead to results close to the 2002 test series (enhanced mixing at the injection) and CFD codes run on fine grid and  $k-\epsilon$  turbulence model are close to the 2004 tests series (injection conditions free of turbulence). For the latter, large differences are obtained between, for example, the two CFX contributions perhaps due to implementation of wall condensation models. Results of GAS2D4 and CFX2D12 match the condensation distribution along the 3 condensers (Phase A - Fig.9) with gas temperature along R4 about 4-5°C above the experimental values and steam concentration relatively close to the experimental results. For Phase B, the contribution of the upper condenser is only matched by GOT2D1, COCLP14 and KUP2D15. The middle condenser part is always overestimated (consequence of the upper condenser contribution and effect of saturation state). Some multi-compartment LP contributions (MELLP2, FUMLP7 and MELLP11) compute dis-

tribution close to the surface ratio between each condenser and the total condensing area (no spatial effect). Apart from ASTLP6, the different ASTEC/COCOSYS contributions lead to almost the same results with different nodalisations.

### 4.3 *Mass and energy balances*

The mass balance is derived from the steady-states, when the injected mass flow is equal to the condensed mass flow. Regarding the energy balance, the extracted power is divided into three components: condensation (latent heat), convection and liquid drain out. Experimentally, about 79% of the power correspond to the latent heat transferred to the wall and 17% is removed by the liquid drain out. The remaining 4% represents the convective part of the heat transfer. Similar distributions were recovered by some participants but it is not possible to correlate the variations with the over or under-estimations of the gas temperature.

### 4.4 *Profiles*

Vertical and radial gas temperature and gas concentration profiles are recovered from the experimental measurements. The first active zone in this simple flow pattern is the rising buoyant jet (injection Richardson number  $Ri = g(\rho - \rho_{inj})D_{inj} / \rho U_{inj}^2$  is 0.09). An example of gas temperature profiles is plotted in Fig. 10. The difference in the two tests series corresponds to the presence of the potential zone above the injection device (5 diameters distance) and the main effect is that hotter gaseous mixture reaches the top of the facility when the potential zone is present (about 5°C difference). Effect of the injection conditions on the jet spreading and the maximum velocity is plotted in Fig. 11. Unfortunately, the quantitative comparison is not possible for the ISP47 Phase A. Instead we use the MICOCO experimental results: 120 g/s of steam injection and 120°C on the three condensers (Blumenfeld and Paillère, 2003). Nevertheless, the 2002 results of the ISP47 have shown quite similar behaviour, ie. the off-centered maximum of the vertical velocity due to misalignment of the laser beams or presence of leakage in the injection device.

Regarding simulation, the LP codes are not designed to simulate this potential zone and the CFD codes with fine grids and k- $\epsilon$  turbulence models reproduce this potential zone. The use of coarse grids or algebraic turbulence models lead to enhanced mixing close to the injection with a faster decrease of the gaseous temperature.

Accurate simulation of this zone may be important for the condensation along the upper condenser (see appendix and Table 3) especially for Phase A steady-state.

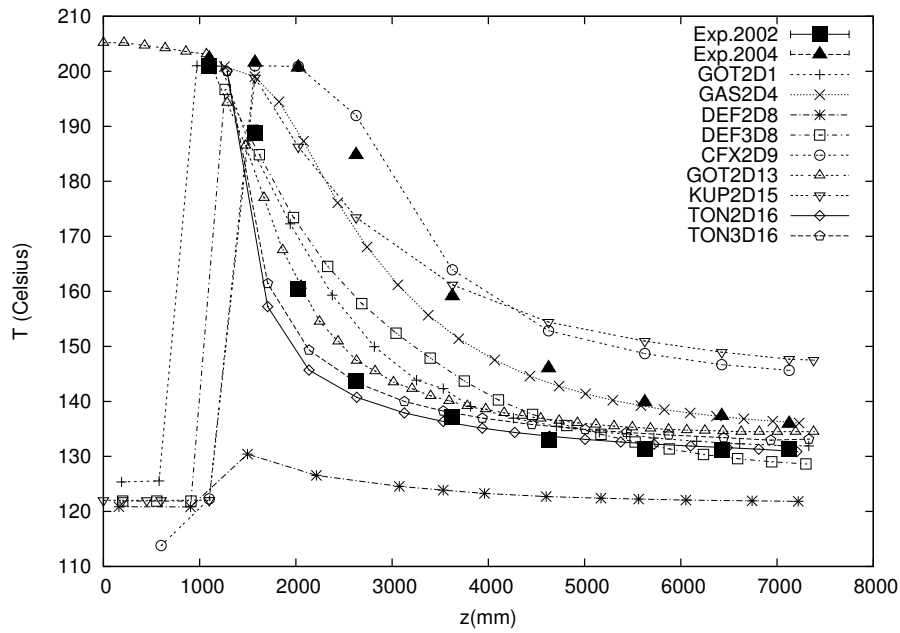


Fig. 10. ISP47 - Phase B - Gas temperature along R0 axis

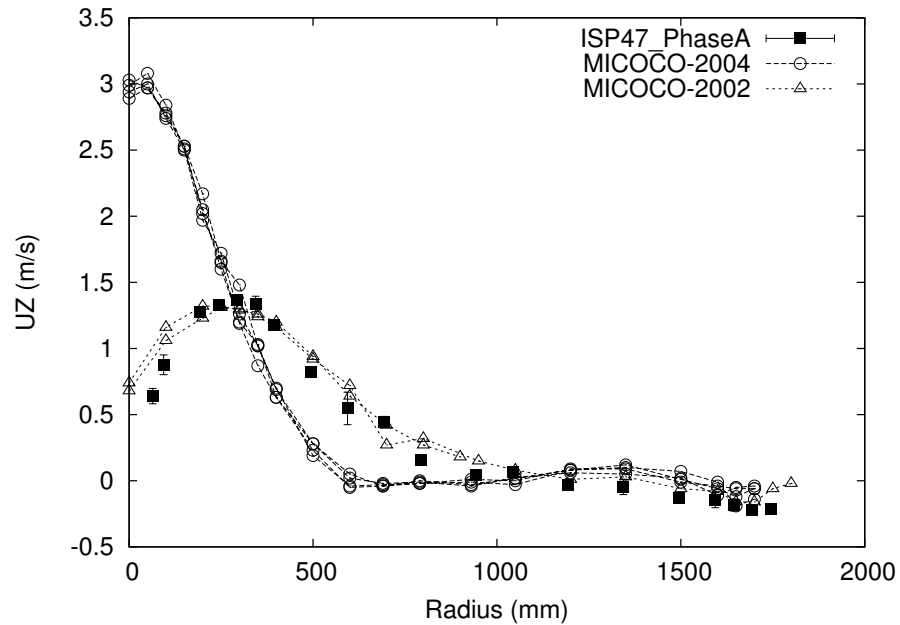


Fig. 11. MISTRA - L.D.V. measurements of the vertical velocity at elevation 4341 mm (MICOCO Air/Steam steady states: 2004 and 2002 tests series compared to ISP47 Phase A 2002 tests series)

The second active zone inside the MISTRA facility is the downward flow close

to the condensers. In this region, due to steam condensation, the gaseous mixture becomes colder and heavier and the non condensable gas fraction increases from the top to the bottom (Fig. 12). Differences are computed in terms of average concentration and gradients. First of all, average values can vary by about 15-20% in relative concentration and this can be correlated with the pressure deviations. Then, gradients vary from less than 1 vol% (MELLP11) up to 6 vol% (KUP2D15) along the height of the three condensers with also opposite sign (KUPLP15 computes an increase in helium content from the bottom to the top). Some computed curves have a constant slope while the others are composed of a two straight lines with different slopes. The experimental measurements show an increase of about 4 vol% in the helium content and due to the measurements uncertainty, it is impossible to determine whether the helium concentration curve has a constant slope or not.

During Phase A steady state (Fig. 13), conditions close to steam saturation are measured and computed along the lower condenser. Fig. 13 shows that superheating increases with height (about 5-7°C superheating at the top of the upper condenser). Some codes predict well the saturation and the superheating, while most codes don't predict saturation but too much superheating. Large differences are computed leading to about 10°C overestimation. Effect of the convective heat transfer coefficient along the condenser has been identified as a key parameter (see appendix) and deeper investigations are needed in this field.

During phase B steady-state (Fig. 14 and 19), conditions close to the steam saturation are reached at the two lower condensers and superheating is limited to the upper condenser. This result is corroborated by the experimental observation through the lower viewing windows (appearance of fog) and the problems related to the LDV measurements. This behaviour is computed by a limited number of codes and it can be clearly seen that LP codes incorporate fog models that constrain saturation as a bounding state and some CFD codes compute supersaturated conditions. Implementation of fog models is a necessary step to deal with such thermodynamic conditions.

The last question regarding gas concentrations concerns the presence or absence of gas stratification in the zero velocity zone around the R2 vertical axis and the achieved mixing between air and helium. The measurements along the R2 axis have shown that the helium concentration difference is in the order of 2 vol% from the top to the bottom of the MISTRA facility (close to homogeneous conditions according to the accuracy of the gas concentration measurement technique). So, we cannot

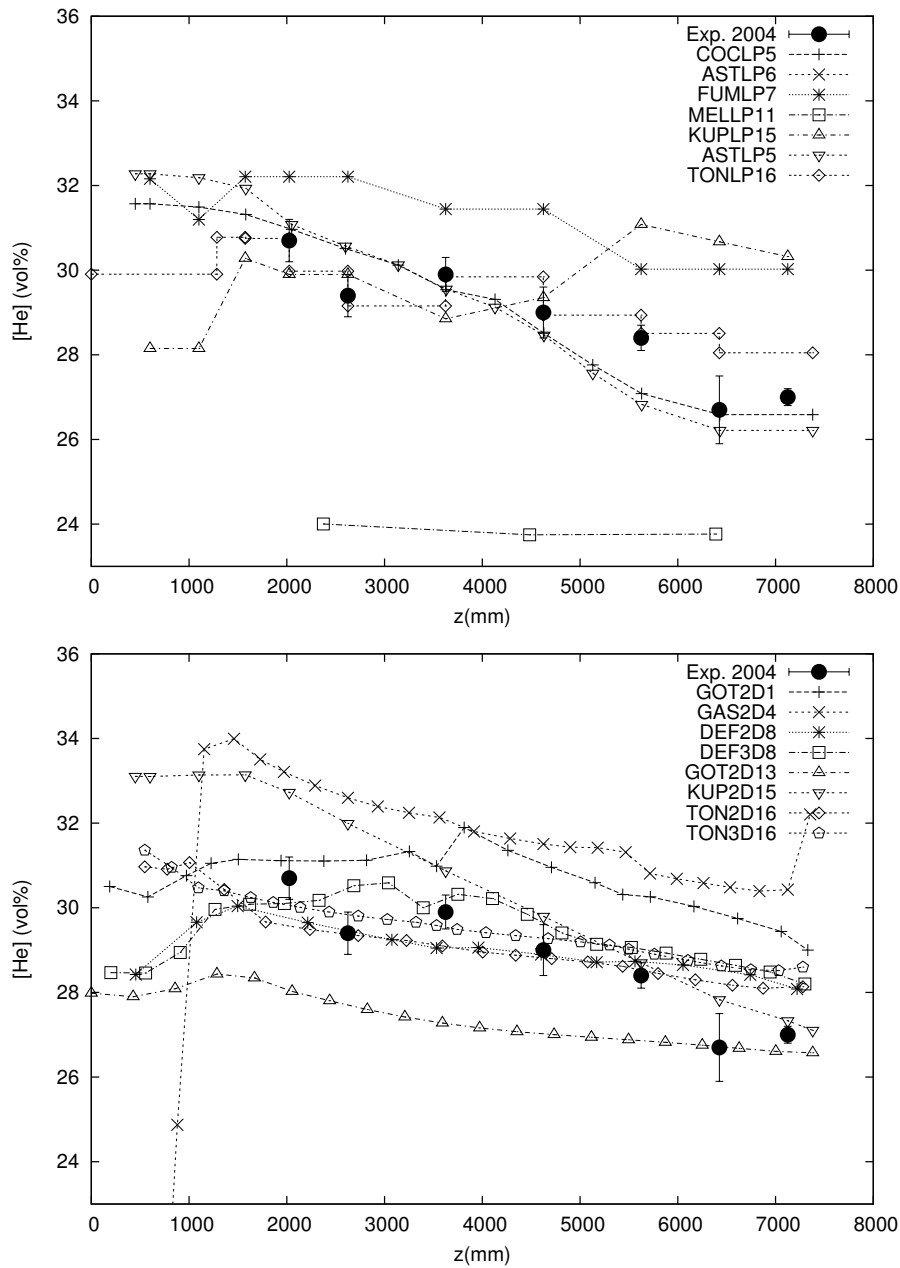


Fig. 12. ISP47 - Phase B - Helium concentration along R4 axis (measurements available only for 2004 tests series)

conclude that there exists any stable gas stratification along the R2 axis or whether the small extent of this zero velocity zone associated to the injection/condensation boundary conditions is not sufficient to provide favourable conditions for a stratification to develop. Using the gas concentration measurements for Phase B steady state, we can conclude that helium is well mixed with the air initially present inside the facility (55.8 vol% of helium in the Air/Helium mixture with a standard deviation of 0.3 vol%). This result is valid in the main gaseous volume (not close

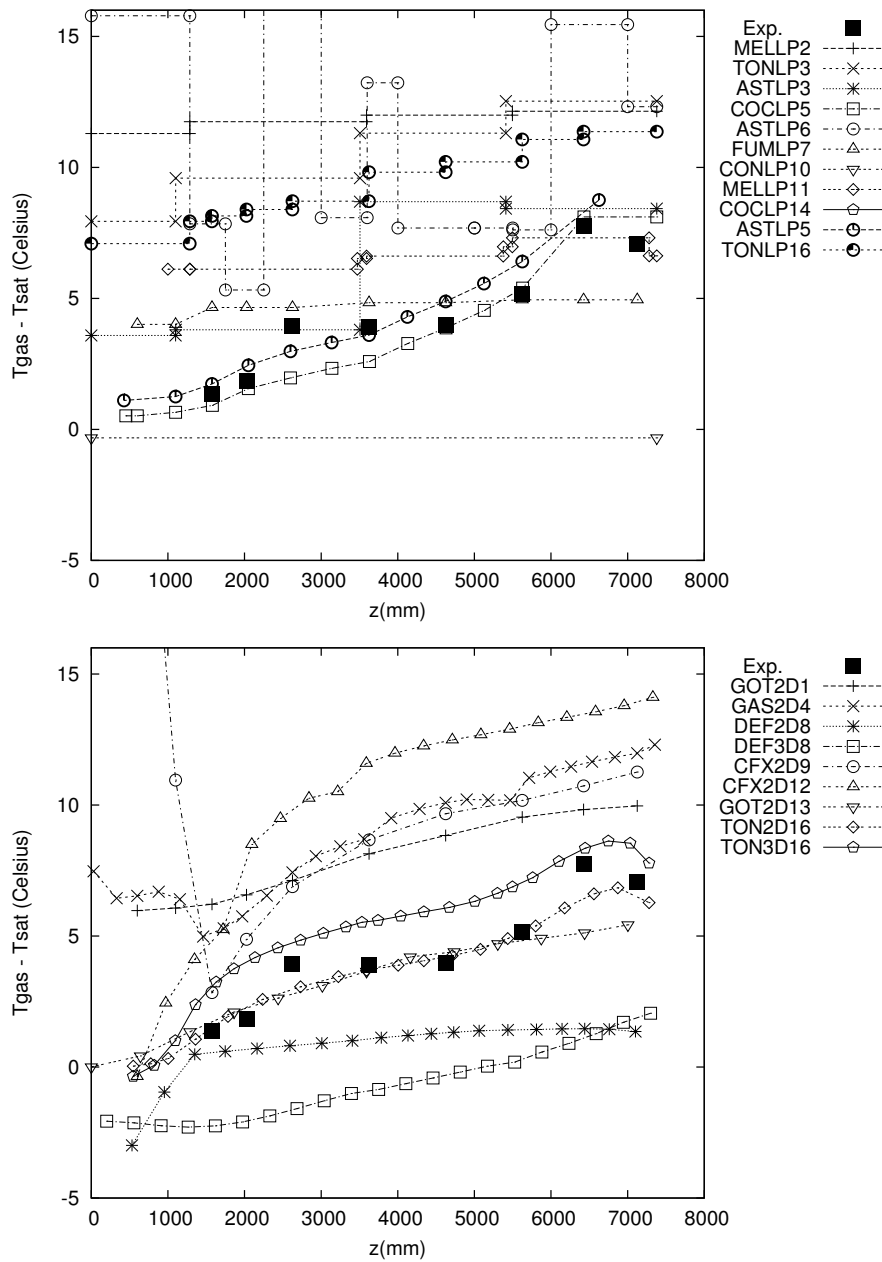


Fig. 13. ISP47 - Phase A - Saturation profile along R4 axis (Experiment: estimations using 2002 experiment 16/04 and confirmed by 2004 tests series - Top: LP codes - Bottom: CFD codes)

to the injection device N5 and N6 levels on R0 axis). Simulations provide the same result.

Finally, typical axial velocity (mean and root mean square value) profiles along the L.D.V. radius ( $z=4341$  mm) are plotted in Fig. 15. The radial grid used by LP codes is too coarse to capture the radial profiles. Nevertheless, the maximum value



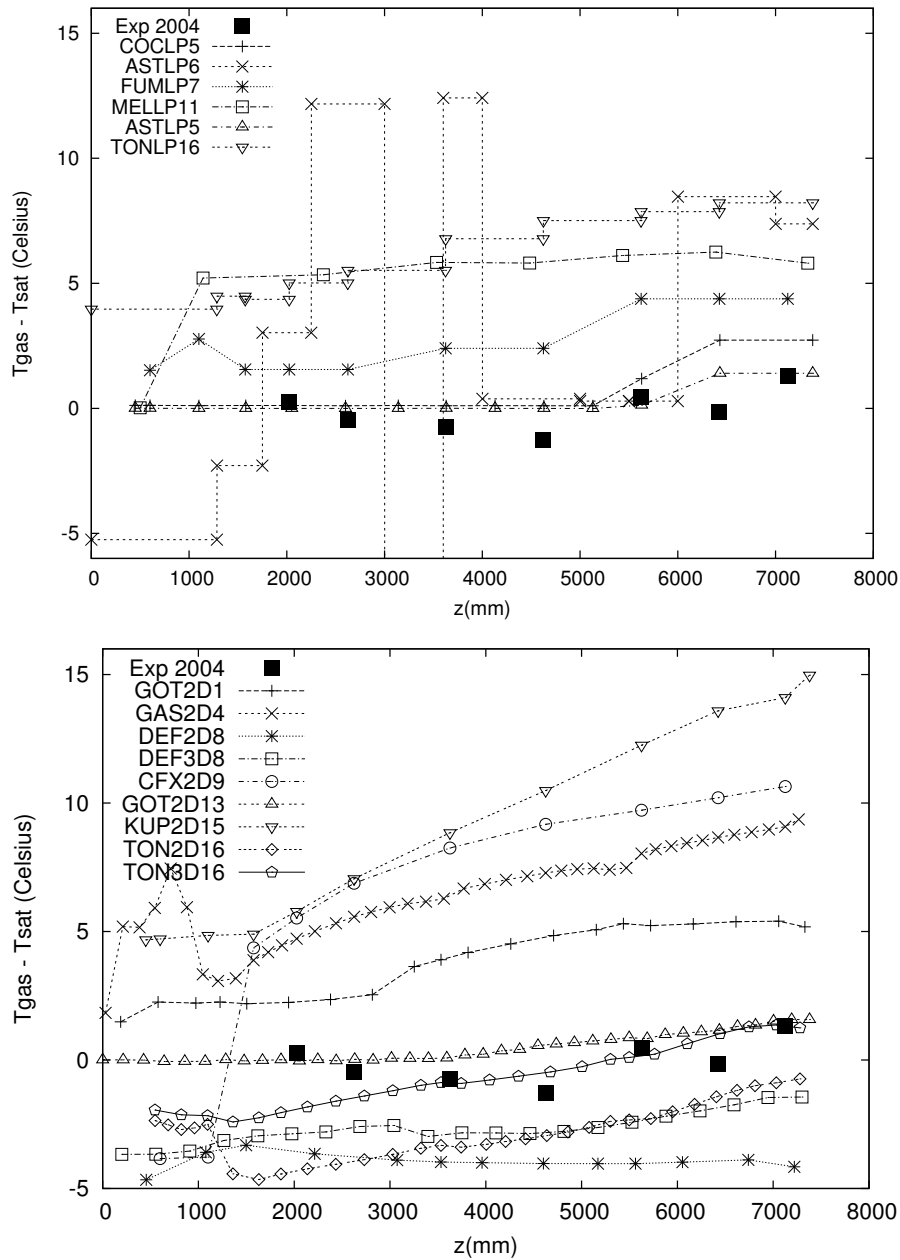


Fig. 14. ISP47 - Phase B - Saturation profile along R4 axis (Top: LP codes - Bottom: CFD codes)

of the axial velocity computed by COCLP5 is comparable to those obtained by the CFD contributions and the “mean” value computed by FUMLP7 or TONLP16 is also in agreement with the CFD ones (between 0 and 1000 mm). For CFD codes, the spread of the maximum value of  $U_z$  is important (between 1 and 2 m/s) and the highest values are related to the contributions that have predicted the potential zone. Then, the jet zone is connected to a zero velocity zone at about 800 mm and the presence of the condensing wall appears after 1400 mm. The computed downwards

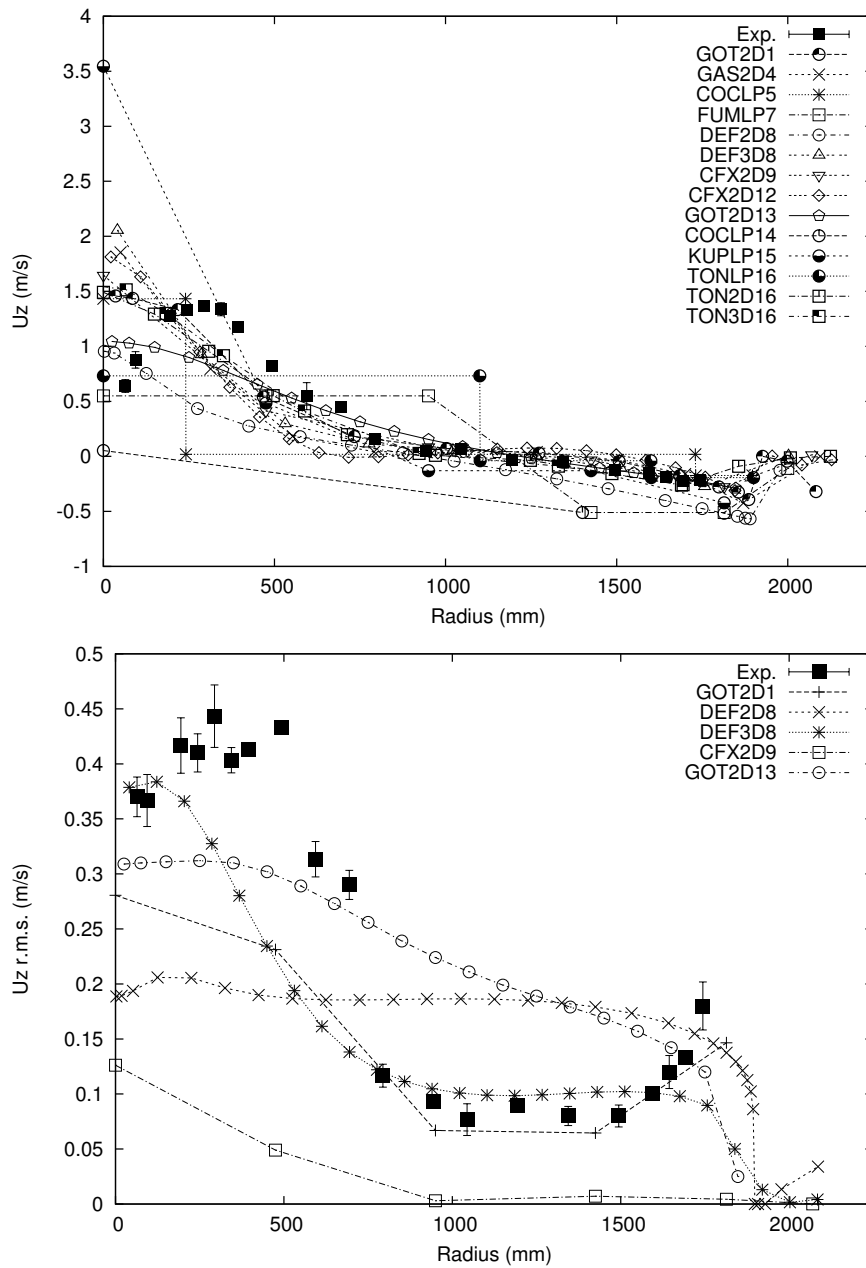


Fig. 15. ISP47 - Phase A - Axial velocity and r.m.s. velocity along L.D.V. axis (Experiment: 2002 tests series)

velocity along the condenser ranges between 0.2 and 0.5 m/s with an experimental value of 0.2 m/s at about 100 mm of the condenser wall. Closer, beam reflexions and low signal to noise values do not allow good measurements. Axial r.m.s velocity profiles clearly identify where the velocity gradients are located: border of the jet and dynamic boundary layer. Simulations explain differences in the radial profiles especially for the GOTHIC contributions.

## 5 Some interesting transient results

Transients are not the first objectives of the MISTRA exercise. Nevertheless, some interesting phenomena have been recorded during and after the transient helium addition to the main steam carrier flow. Along the lower condenser (Fig. 16 and 5), a sudden decrease of the condensation mass flow rate is observed at the beginning of helium addition (4 g/s). At the end of Phase A, the gaseous mixture close to the lower condenser is at the saturation state (Fig. 13) and any change of the mixture composition affect the condensation mass flow rate. Then, a linear decrease of 8 g/s is measured during the helium addition. On the middle condenser, the same decrease is measured without the initial jump and on the upper condenser (Fig. 17), the condensation mass flow rate is not affected by the helium addition. It has been confirmed that this result is independent of the injection temperature (Fig. 4) and injection conditions as such, is the consequence of helium addition. None of the participants really match the above transient behaviour for the three condensers and it seems that the better contribution is coming from the GOTHIC code (MDML model for the lower and middle condenser and global correlation for the upper condenser).

In the experiments, the phase B steady-state is reached after 4 hours of transients and this relatively long time scale is related to the mixing process in the lower head of the MISTRA facility (Fig. 18). In this region, out of the main convective loop, the time scale for helium enrichment (by convection/diffusion processes) corresponds to these 4 hours. The accurate behaviour of the lower head of the facility is not simulated by the participants mainly because attention has not been paid to model this region apart from the main gaseous volume. Post-test calculations regarding this specific point may be interesting and helium enrichment in the lower part of a vessel is investigated in the Thai facility using upper injection with low momentum.

## 6 Discussions and conclusions

First of all, the MISTRA experimental results show it is possible to achieve repeatable measurements on the following phenomena:

- Interaction between injection and wall condensation: modifications in the in-

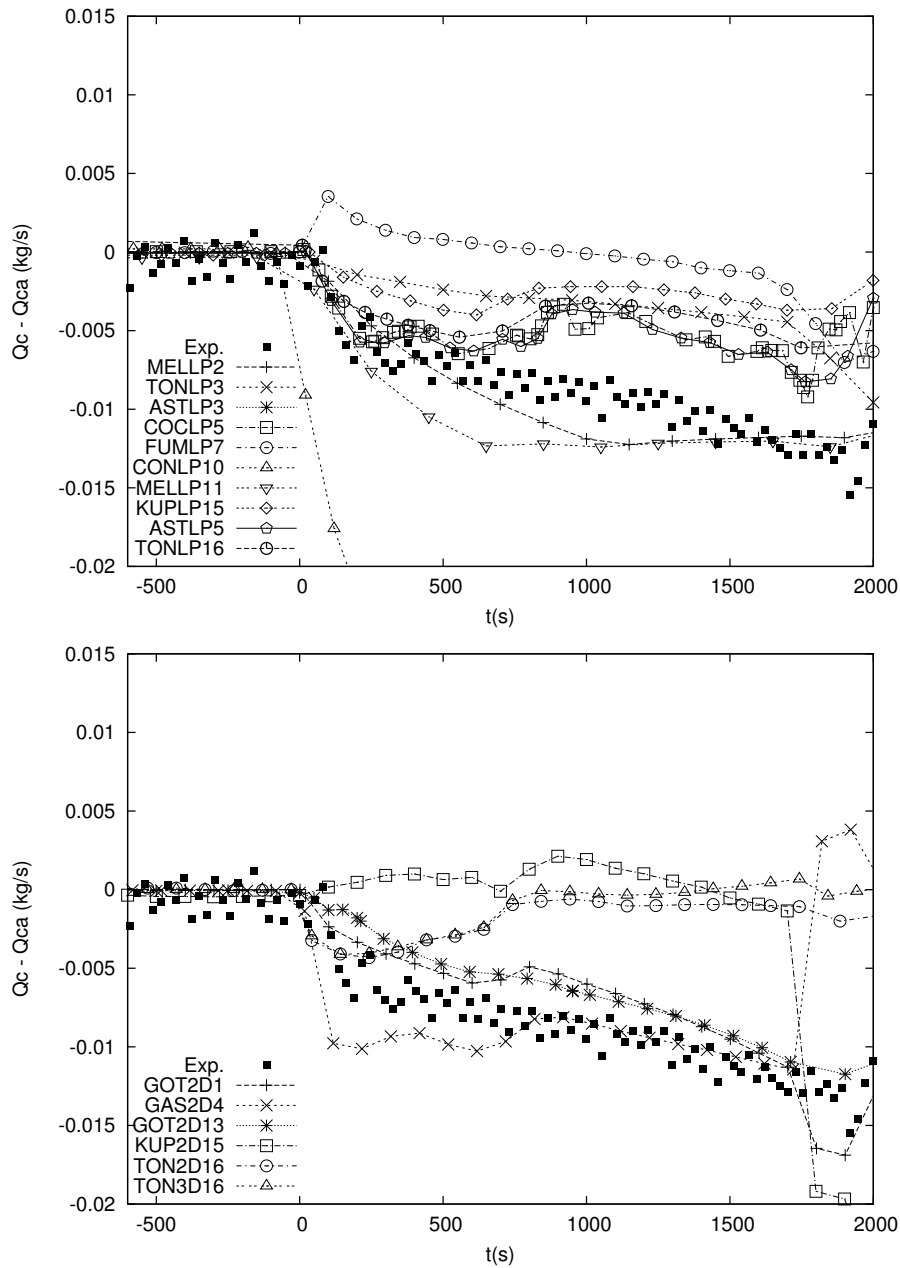


Fig. 16. Transient behaviour of steam condensation along the lower condenser (time zero corresponds to the beginning of helium addition and the results are scaled with Phase A results)

jection (suppression of leakage) change the condensation distribution along the condensers for the two steady-state situations;

- Saturation behavior: helium addition modifies the saturation profile along the three condensers;
- Transient effect of helium addition on the wall condensation.

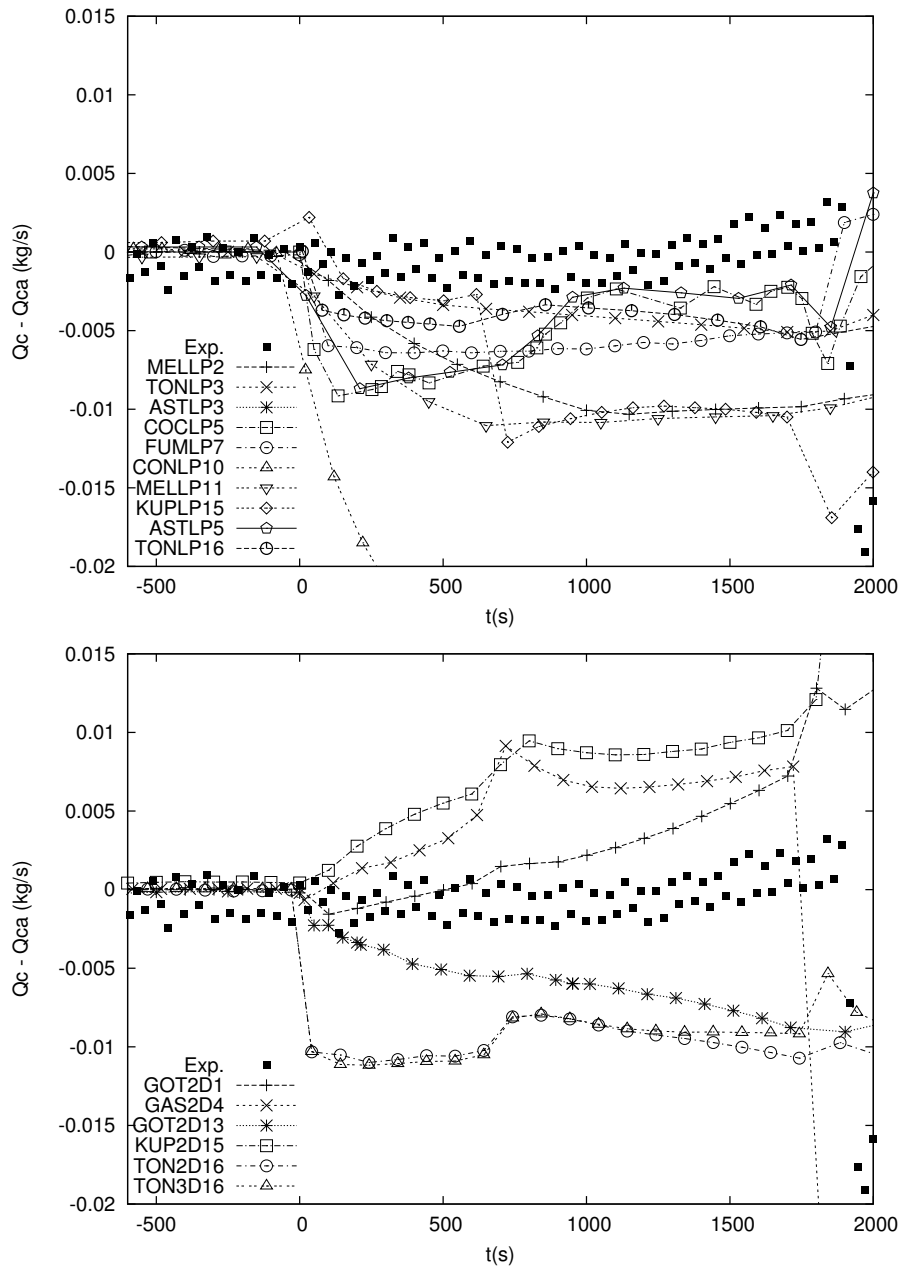


Fig. 17. Transient behaviour of steam condensation along the upper condenser (time zero corresponds to the beginning of helium addition and the results are scaled with Phase A results)

Gas concentration stratification occurs in the lower part of the facility where only few measurement points are available. Also less attention has been paid by the benchmark participants (LP and CFD codes) to this region, and it is not possible to conclude on predictive capabilities of codes regarding gas concentration stratification by using only MISTRA results.

From the blind benchmark exercise, it can be concluded that LP codes give reasonable results if the nodalisation is sufficient to capture the main findings of the flow pattern. Some LP codes are able to simulate the flow pattern when free volumes are subdivided by using adequate nodalisation. Missing user experience can lead to large deviations from experimental values (wrong flow pattern and large pressure overestimation). Guidelines and nodalisation rules must be strictly followed. LP codes usually incorporate fog modeling, and this is an important point for the Phase B steady-state situations in order to avoid subcooled conditions.

Regarding CFD contributions, some capabilities have been demonstrated especially regarding computation of transients, modeling of filmwise steam condensation and jet injection. Experienced users of codes such as GOTHIC or GASFLOW for example provide valuable contributions. Nevertheless some open questions remain such as:

- Simulation of a rising jet with a coarse grid or using an algebraic turbulence model;
- Impact of wall thermal boundary conditions instead of computing the thermal behavior of the steel vessel wall.

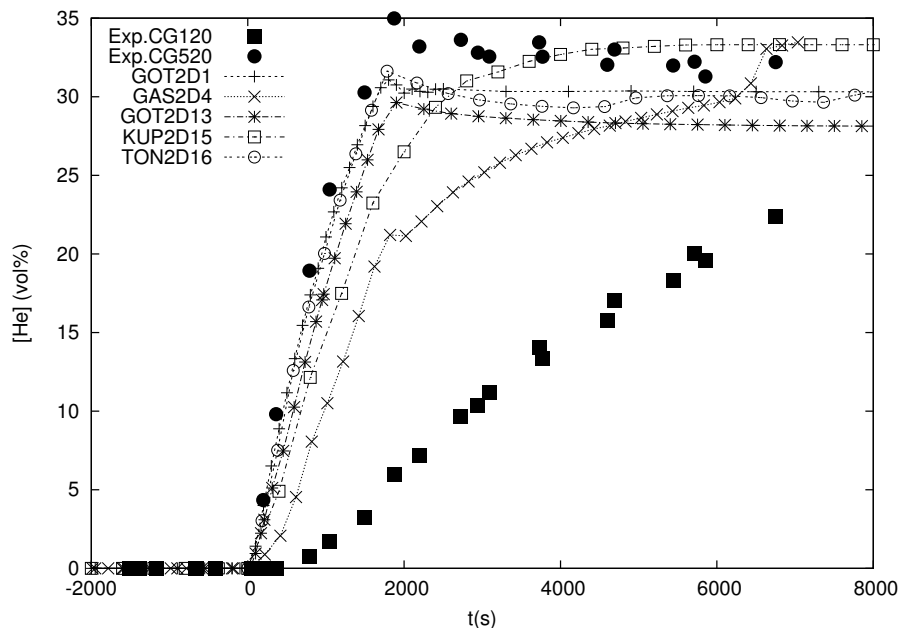


Fig. 18. Transient behaviour of helium addition in the lower part of MISTRA vessel ie below the injection level (time zero corresponds to the beginning of helium addition - Experiment: 2004 tests series)

Implementation of condensation models in a commercial CFD code has been performed by some contributors, but the calculated pressure and temperature data show some notable differences when compared to Phase A results, such that additional development and validation work will be needed. At the time of the MISTRA exercise, only few CFD codes had implemented bulk condensation models; additional improvement and also validation work is needed in this field.

The distribution of the condensation flow along the three condensers is matched by none of the participants for the two steady-state situations and also the transient phase. Differences between measured and calculated data increase for the Phase B results as compared to those of Phase A. Experimental data show that this distribution is affected by the steady-state steam injection conditions and the transient helium addition. Due to interacting phenomena, analysis and explanation of the condensation distribution are difficult, and post-test calculations or simplified analytical case studies (flow impingement for example) may be interesting to enhance the current knowledge.

As also pointed out during former ISP exercise, the use of a computer code (LP or CFD) for containment thermal-hydraulic must be made according to the respect of suitable code guidelines to guarantee an acceptable level of confidence.

Despite relative simple flow pattern inside the MISTRA facility, large deviations (up to 20%) are obtained for the gas temperature and the gas concentration profiles mainly due to an over prediction of the superheating. This may be not compatible with the narrow gas concentration range related to hydrogen combustion processes.

## 7 NOMENCLATURE

$a_i$	: linear response model coefficients
$C_p$	: heat capacity (J/kg/K)
$H$	: heat exchange coefficient (W/m <sup>2</sup> /K)
$k_m$	: mass transfer coefficient (m/s)
$L_{mix}$	: mixing length (m)
$L_v$	: latent heat (J/kg)
$M$	: mass (kg)
$\Phi$	: mass flux (kg/s/m <sup>2</sup> )
$Q$	: mass flow rate (kg/s)
$\rho$	: density (kg/m <sup>3</sup> )
$T$	: temperature (K)
$U$	: velocity (m/s)
$X$	: molar fraction
$X_i$	: parameters
$Y_i$	: responses
$inj$	: subscript for injection conditions
r.m.s.	: root mean square

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## **APPENDIX: Parametric study using Design-Of-Experiments techniques around Phase A steady-state using TONUS CFD code**

Effect of the user parameters and the experimental uncertainty on the boundary conditions have to be quantified to provide a complete numerical study of this ISP exercise. The first step is to select the parameters and the range of each one. The second step is to define the variables to study. In the following paragraphs, these variables are called responses. Then, a methodology to perform the sensitivity study has to be chosen. In our case, the Design-Of-Experiments techniques have been applied in order to optimise the number of calculations (mainly due to CPU and memory costs with CFD code). Finally, the results have to be analysed.

The parameters may have different origins. The first one comes from the user parameter included in each computer code. Five parameters have been identified in the present TONUS modelling. One is related to the turbulence modelling: the mixing length ( $L_{mix}$ ). Two are related to heat transfer coefficient: convective heat transfer along the condensers ("user" specified)  $H_{conv}$  and the external heat transfer coefficient  $H_{ext}$  (used to simulate the heat losses) and the two last are related to the perfect gases hypothesis used to model the steam: the steam heat capacity  $Cp_{steam}$  and the latent heat  $L_v$ . The second origin is the uncertainties in the experimental boundary conditions. The six selected parameters are related to the initial conditions in the MISTRA facility (initial mass of air  $M_{air}$ ), the injection steam mass flow rate ( $Q_{steam}$ ), injection temperature ( $T_{inj}$ ) and the surface temperature of the three condensers ( $T_{c_{up}}$ ,  $T_{c_{med}}$  and  $T_{c_{low}}$ ). The variations chosen for all these parameters are summarised in Tab. 4 and they correspond to estimated experimental uncertainty or engineer's judgement of reasonable variations.

Parameter	Name	Variation (+/-)
Initial Mass of Air	$M_{air}$	1,6%
Injection Steam Mass Flow Rate	$Q_{steam}$	3 g/s
Injection Temperature	$T_{inj}$	4°C
Upper Condenser Temperature	$T_{Cup}$	1°C
Middle Condenser Temperature	$T_{Cmed}$	1°C
Lower Condenser Temperature	$T_{Clow}$	1°C
Mixing Length	$L_{mix}$	50%
External Heat Exchange Coefficient	$H_{ext}$	2W/m <sup>2</sup> /K
Convective Heat Transfer Coefficient along the condensers	$H_{conv}$	40%
Steam Heat Capacity	$Cp_{steam}$	6%
Latent Heat	$L_v$	4%

Table 4

Parameters selected for the sensitivity study

$$A = \begin{pmatrix} -1 & -1 & -1 & -1 & -1 & -1 & -1 & -1 & -1 & -1 & -1 \\ -1 & -1 & -1 & -1 & -1 & 1 & 1 & 1 & 1 & 1 & 1 \\ -1 & -1 & 1 & 1 & 1 & -1 & -1 & -1 & 1 & 1 & 1 \\ -1 & 1 & -1 & 1 & 1 & -1 & 1 & 1 & -1 & -1 & 1 \\ -1 & 1 & 1 & -1 & 1 & 1 & -1 & 1 & -1 & 1 & -1 \\ -1 & 1 & 1 & 1 & -1 & 1 & 1 & -1 & 1 & -1 & -1 \\ 1 & -1 & 1 & 1 & -1 & -1 & 1 & 1 & -1 & 1 & -1 \\ 1 & -1 & 1 & -1 & 1 & 1 & 1 & -1 & -1 & -1 & 1 \\ 1 & -1 & -1 & 1 & 1 & 1 & -1 & 1 & 1 & -1 & -1 \\ 1 & 1 & 1 & -1 & -1 & -1 & -1 & 1 & 1 & -1 & 1 \\ 1 & 1 & -1 & 1 & -1 & 1 & -1 & -1 & -1 & 1 & 1 \\ 1 & 1 & -1 & -1 & 1 & -1 & 1 & -1 & 1 & 1 & -1 \end{pmatrix} \quad (2)$$

So, eleven parameters have been selected for this sensitivity study. To avoid a very large number of calculations ( $2^{11} = 2048$ ), it has been decided to apply Design-Of-Experiments techniques (Lepallec et al., 2003). A screening approach has been adopted because at present time, only the first order is of interest and so a linear response model (LRM) without interactions has been selected. If a result is called  $Y$  (response) and the parameters  $X_i$  for  $i=1$  to 11, 12 coefficients of the linear response models have to be determined called  $a_i$  for  $i=0$  to 11.

$$Y_i = \sum_{i=0}^{11} a_i X_i \quad (3)$$

This needs 12 calculations using the calculation matrix  $A$  (see eq. 2) where each row corresponds to a calculation and each column to a parameter  $X_i$ . The value of the parameters have been normalised in order to allow direct comparison of their value. The accuracy of the Linear Response can be checked at the end by a direct comparison of the model response in the centre of the variation domain compared to a calculation with the parameter at their reference value. The sensitivity study has been performed with the TONUS CFD using the 2D axisymmetric model using only the Phase A boundary conditions. Absence of accurate fog modelling in the TONUS code does not allow the same study for the Phase B steady-state. This will be rerun when the model is available.

Several responses ( $Y_i$ ) have been selected including the pressure ( $P_{tot}$ ), mean gas temperature ( $T_{fm}$ ), condensation mass flow rate on the three condensers ( $Q_c$ ), spurious condensation ( $Q_{c,spurious}$ ), the temperature difference between levels N3 and N10 on R2 axis ( $\Delta T_{h/b}$ ), difference between  $T_{gas}$  and  $T_{sat}$  at level N3 and N10 on R2 axis and finally, the maximum and the minimum vertical velocity at the LDV radius. The results of the sensitivity study are reported in Table 5. The different coefficients in the main table ( $a_1$  to  $a_{11}$ ) are expressed in terms of % of the selected response  $Y_i$ . For example, setting the external heat exchange coefficient ( $H_{ext}$ ) to its highest estimated value (+1 or  $6W/m^2/K$ ) leads to an increase (+ sign) of the spurious condensation mass flow rate of about 50.8%. Before interpreting this result, it is interesting to check the validity of the LRM by comparing the results of the LRM in the centre of the variation domain ( $a_0$  coefficient) with the corresponding 2D calculation (called 2D verification). The errors for the global variables are quite satisfactory (below 5%). The maximum values are related to the condensation mass flow rates on the three condensers and these responses are strongly coupled together so interactions between parameters have to be taken into account if the goal is to obtain more accurate results. This remark is also valid for most of the lo-

	$P_{tot}$ (bars)	$T_{fm}$ (°C)	$Q_{c,low}$ (g/s)	$Q_{c,med}$ (g/s)	$Q_{c,up}$ (g/s)	$Q_{c,spurious}$ (g/s)	$\Delta T_{h/b}$ (°C)	$\Delta T_{sat,h}$ (°C)	$\Delta T_{sat,b}$ (°C)	$U_{z,max}$ (m/s)	$U_{z,min}$ (m/s)
Mean $a_0$ (1)	3.314	124.96	30.95	34.88	42.86	21.28	8.49	3.07	9.25	1.68	-0.26
$a_1, M_{air}$	<b>1.44</b>	0.03	-0.23	-0.63	-0.05	0.69	-4.39	6.95	-1.73	0.29	-0.57
$a_2, Q_{steam}$	0.18	0.11	4.16	2.80	2.26	-0.89	3.42	-3.58	1.57	1.14	1.20
$a_3, T_{inj}$	-0.03	0.31	-0.35	0.68	-0.24	0.05	5.76	6.89	7.20	1.80	1.70
$a_4, T_{c,up}$	0.56	0.23	7.67	8.31	<b>-12.87</b>	0.87	-3.33	7.11	-0.36	-0.66	-0.76
$a_5, T_{c,med}$	0.45	0.21	9.87	<b>-15.78</b>	5.60	1.03	2.58	-4.51	0.27	-0.88	-0.76
$a_6, T_{c,low}$	0.56	0.28	<b>-17.75</b>	7.29	6.94	1.02	0.64	0.98	1.17	-0.80	0.51
$a_7, L_{mix}$	-0.40	-0.50	7.72	-0.34	-4.90	-1.34	<b>-30.03</b>	21.66	-12.41	<b>-26.31</b>	<b>7.77</b>
$a_8, H_{ext}$	<b>-1.07</b>	-0.28	-10.41	-9.83	-9.41	<b>50.80</b>	2.03	4.45	2.95	0.48	0.25
$a_9, H_{conv}$	0.00	<b>-1.67</b>	-1.59	0.45	-0.56	1.20	-2.36	<b>-76.22</b>	<b>-27.05</b>	2.00	1.89
$a_{10}, Cp_{steam}$	0.03	0.32	0.20	-0.12	0.48	0.01	3.23	8.96	5.75	-0.35	0.69
$a_{11}, L_v$	0.15	0.02	0.42	0.54	0.90	-3.87	-3.52	4.83	-0.77	1.35	0.82
Ref. (2)	3.303	124.07	32.66	33.69	41.99	21.65	7.57	2.64	8.20	1.49	-0.26
Err. (1-2)/2	0.003	0.007	-0.052	0.035	0.021	-0.017	0.122	0.163	0.128	0.129	0.010

Table 5

Results of the sensitivity study (bold font is used to highlight the main contributors and Ref. corresponds to the 2D verification of the linear response model)

cal variables because the errors are greater than 12% except the minimum velocity for which the linear response model is as accurate as for the global variables. In a screening approach, these results are sufficient for an interesting interpretation of

the results.

Pressure mainly depends on the initial mass of air and the accurate simulation of the heat losses. Then, parameters acting on condensation (temperature of the condenser) are important.

Mean gas temperature is mainly driven by the convective heat exchange along the condensers (energy transfer without mass). Then, the mixing length acting on the mixing process of the rising jet becomes important and finally, injection conditions follow.

Condensation mass flow rate is first controlled by the temperature of the condensers. Then, the impact of the spurious condensation is evident (controlled by  $H_{ext}$ ) and finally, the mixing length becomes an important parameter (decrease of the mixing length will decrease the mixing process and enhanced the condensation mass flow rate along the upper condenser). This last parameter has different impact on the different condensers.

Gas temperature difference along R2 axis is mainly dependent on the mixing hypothesis (choice of the mixing length). Degree of saturation along R2 axis is first controlled by the convective heat transfer coefficient along the condensers (increase of the heat transfer coefficient will lead to conditions closer to the saturation state). Velocities are controlled as expected by the choice of the mixing length. The impact is more important for the jet zone instead of the condensation zone. These results provide some explanations of the experimental and computational results and this may give some guidelines to investigate some post tests calculations. Such studies are a powerful tool for interpretation.

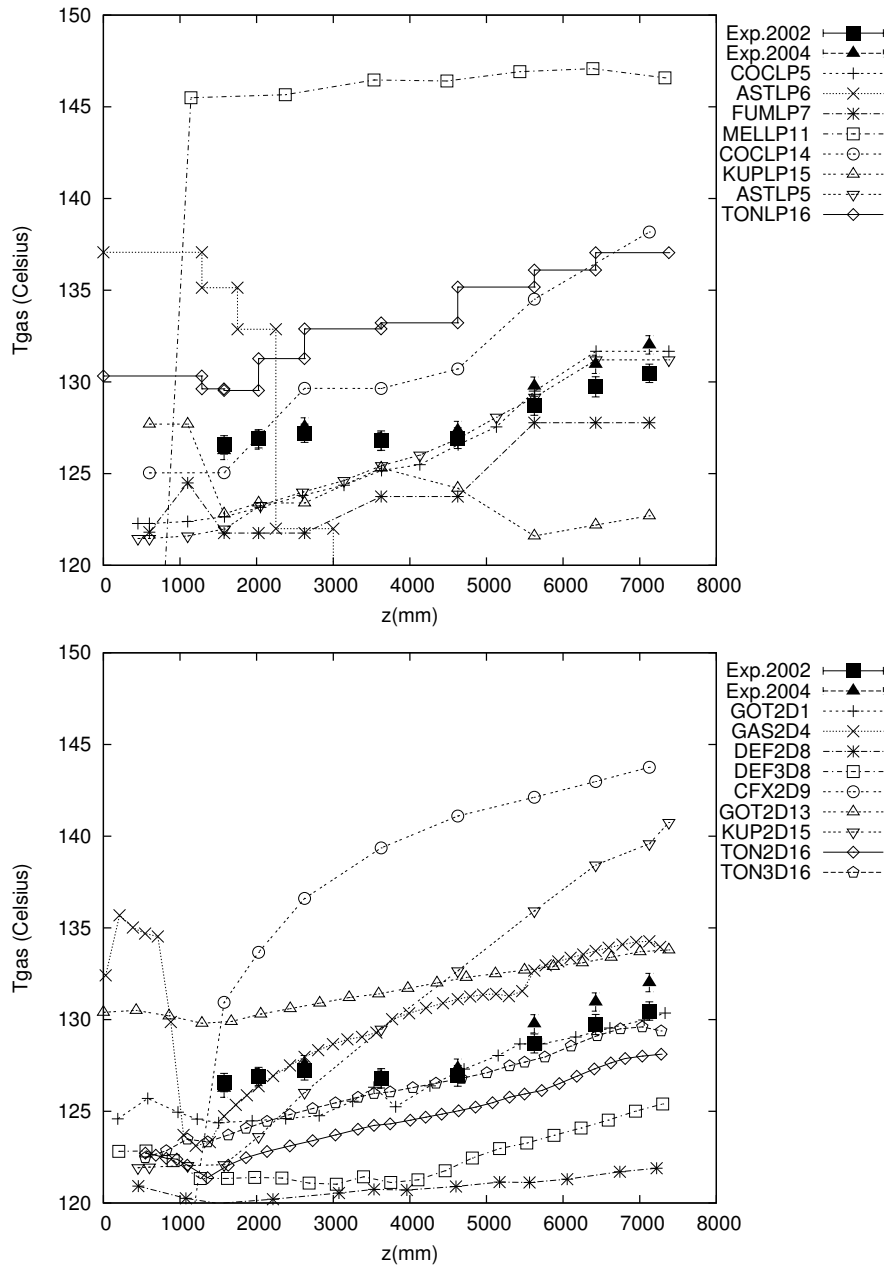


Fig. 19. ISP47 - Phase B - Gas temperature profile along R4 axis (Top: LP codes - Bottom: CFD codes)